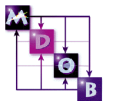


# On Using Approximations in Engineering Optimization

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# Outline

- **Background**
  - Motivation
  - Some Recent Developments in Approximations for Engineering
- **Model Trust-Region Approach with General Approximations**
  - The Model-Management Framework
  - Convergence Analysis
  - Example (Eddy-Promoter Heat Exchanger)
- **Current Research**
  - Constraints and MDO
  - Novel Applications (HSCT, Aerospike, Rotorcraft)
- **Summary**



# Approximations-in-Optimization Problem

- The problem:

$$\text{minimize } f(x)$$

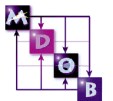
The essential problem is multidisciplinary and has special structure, but here will consider single discipline, unconstrained optimization.

- Motivation

- Address computational expense issues of using high-fidelity approximation models in optimization (Example: Navier-Stokes vs. Euler)
- Allow for easier integration of disciplines in multidisciplinary context
- Allow for interactive design

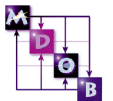
- Some history

- Schmit, et al. - First explicit coupling of structural analysis and nonlinear programming (1960); Some approximation concepts for structural synthesis (1960)
- Fleury, et al. (1989) - Approximation strategies in structural optimization (analysis)
- Barthelemy, et al. (1993) - Overview of approximation concepts in structural design
- ...



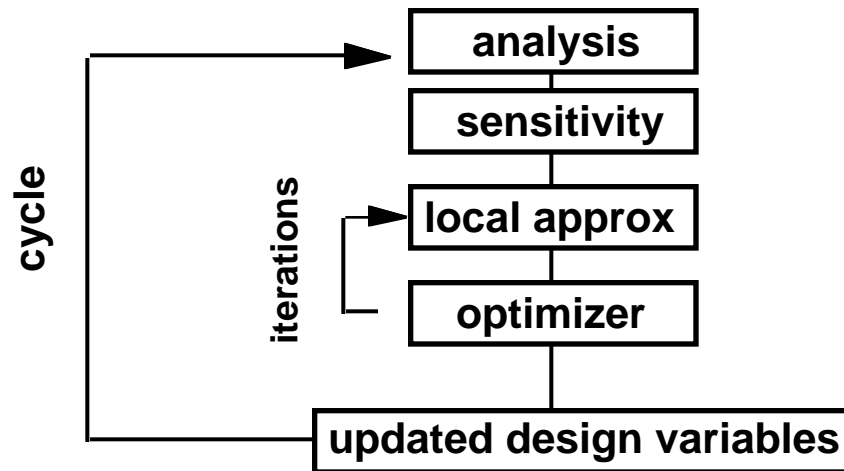
# Approximations-in-Optimization Problem (cont.)

- **Existing practices**
  - Use a variety of fidelities for models or approximations managed via heuristics
  - Examples: physical models, statistical models, move limits
- **Difficulties with heuristics**
  - There is no guarantee that a design that promises improvement with a low-fidelity system will yield improvement in the high-fidelity system
  - It is not clear when to refine the model
  - Robustness is not assured



# Existing Practices: Example

Optimizer and approximate analysis optimization scheme for HSCT (Walsh, et al.)

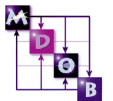


- Evaluate objective, constraints, and derivatives of objectives and constraints at the beginning of cycle;  $f_0$  is a coupled Navier-Stokes and finite-element analysis
- During optimization iterations,  
do ...  
    call optimizer with  $f = f_0 + \frac{f_0}{x} \cdot x$  (similarly for constraints  $g$ )  
    + move limits  
end do



# Some Recent Developments in Engineering Approximations

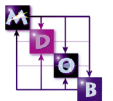
- Research conducted or supported at NASA Langley:
  - Design-oriented analysis and approximations
    - E. g., at University of Florida / Virginia Tech (Haftka, et al.)  
(Response Surface Approximations in High-dimensional Spaces Using Several Levels of Fidelity)
  - Approximation / modeling validation
    - E. g., at MIT / NASA Langley (Otto, et al.)  
(Computer Simulation Surrogates for Numerical Simulations and Optimization; Surrogate Pareto Approach to Shape Optimization)
  - Managing approximation models in optimization
    - ➡ • E. g., at **NASA Langley / ICASE;**  
Notre Dame / Virginia Tech (Rodriguez, et al.)  
(Augmented Lagrangian Response Surface Approximations - Model Management Framework for General Constrained Optimization)
- Links to detailed information provided at:  
<http://fmad-www.larc.nasa.gov/mdob>



# A Trust-Region Framework for Managing the Use of Approximation Models in Optimization

(Results by Alexandrov, et al. in *Journal on Structural Optimization*)

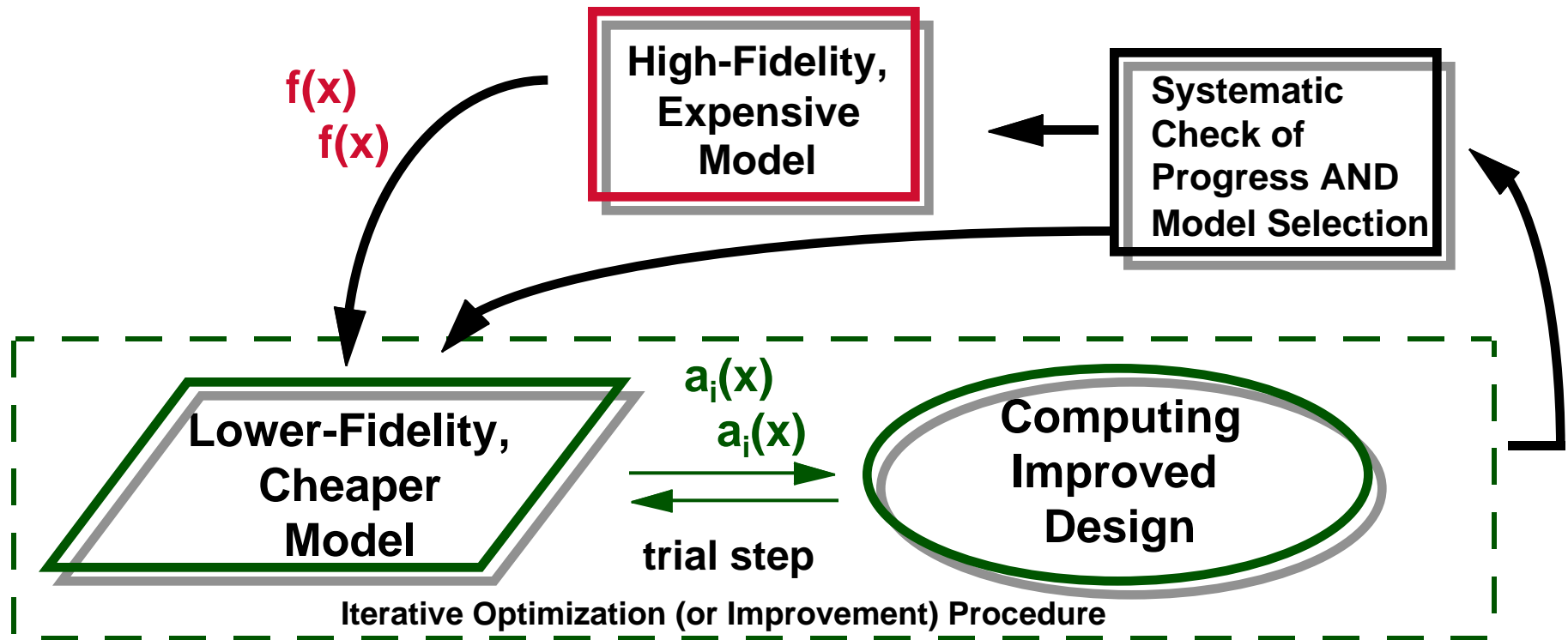
- This research considers general first order approximations and answers the question “How does one make an approximation scheme robust”, in particular:
  - What does one do when the design derived with a lower-fidelity model fails to produce improvement in the true objective?
  - How does one use information about the predictive value of an approximation to adjust the amount of optimization with a lower-fidelity model before recourse to the higher-fidelity model?
  - How does one use approximations to yield an answer to the high-fidelity problem?
- Observation:
  - The framework provides a method for managing the use of models of varying **physical** fidelity



# Model Trust-Region Approach with General Approximations

$f(x)$  - high fidelity, expensive model, such as analysis or simulation

$a_i(x)$  - one of the suite of lower fidelity or accuracy models of the same physical process





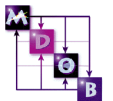
# Requirements on the Approximation Model

- At each major point  $x_k$ , the following model consistency conditions are assumed to hold:

$$a_k(x_k) = f(x_k)$$

$$a_k(x_k) = f(x_k)$$

- **Observations:**
  - Consistency is only enforced at the “anchor” points.
  - The gradients do not have to match exactly, but that is the assumption made in the published paper



# Requirements on the Approximation Model: Enforcing the Consistency Conditions

- In practice, consistency is an application dependent question, but there exist methods for enforcing consistency.
- Example: Correction by  $\gamma$ -correlation approach. Chang et al. (1993)

Assuming no specific functional form, let  $f_{lo}$  be a model of lower fidelity than  $f$ . Define

$$\gamma(x) = \frac{f(x)}{f_{lo}(x)}$$

Given  $x_k$ , build

$$\gamma_k(x) = \gamma(x_k) + (\gamma(x_k))^T (x - x_k)$$

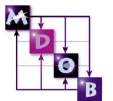
Use  $\gamma_k$  to scale the lower-fidelity model  $f_{lo}$ :

$$f(x) = \gamma_k(x) a(x) \gamma_k(x) f_{lo}(x)$$

Then

$$a_k(x) = \gamma_k(x) f_{lo}(x)$$

satisfies the consistency conditions.



# The Algorithm with General Approximations

Choose  $x_0 \in \mathbb{R}^n$ ,  $\epsilon_0 > 0$

For  $k = 0, 1, \dots$  until convergence do

Choose  $a_k$  such that  $a_k(x_k) = f(x_k)$  and  $a_k(x_k + s) \approx f(x_k + s)$

Compute an approximate solution  $s_k$  to subproblem:

$$\begin{aligned} &\text{minimize } a_k(x_k + s) \\ &\text{subject to } \|s\| \leq \epsilon_k \end{aligned}$$

Compare the actual and predicted decrease in  $f$ :

$$r = \frac{f(x_k) - f(x_k + s_k)}{f(x_k) - a_k(x_k + s_k)}$$

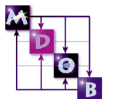
$$x_{k+1} = \begin{cases} x_k + s_k & \text{if } f(x_k + s_k) < f(x_k) \\ x_k & \text{otherwise} \end{cases}$$

and

$$\epsilon_{k+1} = \begin{cases} c_1 \epsilon_k & \text{if } r < r_1 \\ \min\{c_2 \epsilon_k, \epsilon_{\max}\} & \text{if } r > r_2 \\ \epsilon_k & \text{otherwise} \end{cases}$$

for some  $r_1 < r_2 < 1$ ,  $c_1 < 1$ ,  $c_2 > 1$

end do



# Convergence Analysis

- **Observations:**

- Practical performance will depend on the quality of the approximation models and their ability to predict the behavior of  $f$ .
- Options in case of unsuccessful step:
  - Improve model fidelity
  - Do less optimization - reduce the trust-region radius.

- **Convergence:**

- We solve approximately:

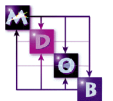
$$\begin{aligned} &\text{minimize } a_k(x_k + s) \\ &\text{subject to } \|s\| \leq \Delta_k \end{aligned}$$

with  $a_k$  - a general 1st order model.

- “Approximately” =  $s_k$  must satisfy Fraction of Cauchy Decrease (FCD). Use the variant: there exist  $\eta$ ,  $C > 0$ , independent of  $k$ , such that  $s_k$  satisfies

$$\|f(x_k) - q_k(x_k + s_k)\| \leq \eta \|f(x_k)\| \min(\Delta_k, \|f(x_k)\|/C).$$

- The following algorithm for solving the subproblem satisfies FCD:



## Computing an approximate solution $s_k$

Given  $x_k \in \mathbb{R}^n$ ,  $\epsilon_k > 0$ , choose  $\alpha \in (0,1)$ ,  $a_1, a_2 > 0$  and set  $y_0 = x_k$ ,  $\epsilon_0 = \epsilon_k$ ,  $v_0 = 0$ .

For  $j = 0, 1, \dots$ , until  $\epsilon_{j+1} < \epsilon_k$  do

Find an approximate solution  $p_j$  to:

$$\begin{aligned} &\text{minimize } q_j(y_j + p) = a_k(y_j) + a_k(y_j)^T p + \frac{1}{2} p^T H_j p \\ &\text{subject to } \|p\| \leq \epsilon_j \\ &\quad \|y_j + p\| \leq \epsilon_k \end{aligned}$$

that satisfies FCD for  $a_k$  from  $y_j$ .

Compare the actual and predicted decrease in  $a_k$ :

$$r = \frac{a_k(y_j) - a_k(y_j + p_j)}{a_k(y_j) - q_j(y_j + p_j)}$$

Update  $y_j$  as indicated below, update  $\epsilon_j$  as  $\epsilon_k$

$$v_{j+1} = v_j + (y_{j+1} - y_j)$$

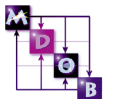
end do

Set  $s_k = v_j$

Updating  $y_j$ : choose  $\mu > 0$ , independent of  $k, j$ , and set

$$\text{If } y_j = x_k, \text{ then } y_{j+1} = \begin{cases} y_j + p_j & \text{if } r > \mu \\ y_j & \text{otherwise} \end{cases}$$

$$\text{If } y_j \neq x_k, \text{ then } y_{j+1} = \begin{cases} y_j + p_j & \text{if } r > 0 \\ y_j & \text{otherwise} \end{cases}$$



# Convergence Analysis

- The subproblem is itself a TR subproblem

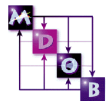
$$\begin{aligned} \text{minimize } q_j(y_j + p) &= a_k(y_j) + a_k(y_j)^T p + 1/2 p^T H_j p \\ \text{subject to } \|p\| &\leq \rho_j \\ \|y_j + p\| &\leq r_k \end{aligned}$$

- Exact and approximate solutions are given in Heinkenschloss (1994).
- Let  $p_N$  be the first acceptable step. It satisfies FCD for  $a_k$  at  $x_k$  and since  $r > \mu$ ,

$$\|a_k(x_k) - a_k(x_k + p_N)\| \leq \mu \|a_k(x_k)\| \min(\rho_N, \|a_k(x_k)\|/C)$$

- Applying the consistency conditions and assuming uniform boundedness in  $k$  of Hessians  $\nabla^2 a_k(x+s)$  for all  $s$  with  $\|s\| \leq \rho_k$  (the latter guarantees the existence of  $\rho$ , independent of  $k$  for which  $\rho_N = \rho$ ) yields:

$$\|f(x_k) - a_k(x_k + p_N)\| \leq \mu \|f(x_k)\| \min(\rho_k, \|f(x_k)\|/C).$$



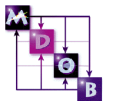
# Convergence Analysis

- Since any steps after  $N$  decrease  $a_k$  further, the step produces FCD for  $a_k$  as an approximation of  $f$  at  $x_k$  and Powell's global convergence theorem (1975) is applicable:
  - If  $f$  is uniformly bounded below, uniformly continuously differentiable, and the Hessian approximations are uniformly bounded, then the sequence of iterates generated by a trust-region algorithm whose steps satisfy FCD satisfies

$$\liminf_{k \rightarrow \infty} \|f(x_k)\| = 0$$

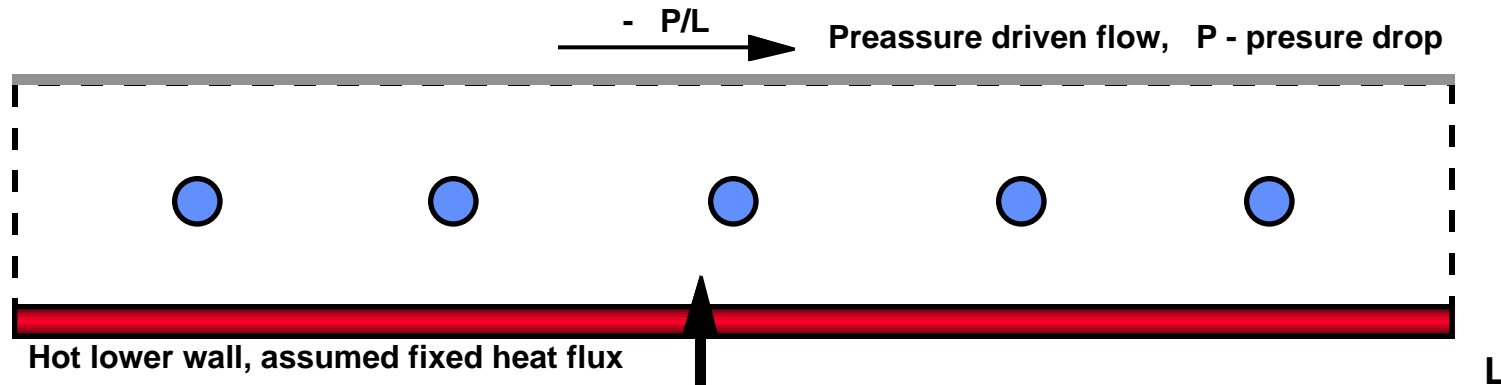
- The use of the acceptance criterion for  $y_j$  further guarantees that

$$\lim_{k \rightarrow \infty} \|f(x_k)\| = 0$$



# Illustrative Example

## “Eddy-Promoter” Heat Exchanger (analyses / codes by Otto et al.)



- **Goal**
  - Transfer heat from the lower surface into the fluid medium as efficiently as possible
- **Objectives**
  - Lower wall temperature
  - Lower preassure gradient
- **Eddy-Promoter Configuration**
  - Periodic array of cylindrical obstructions
  - Lowers critical parameter for onset of instability
  - Improved heat transport
- **Governing Equations**
  - 2-D incompressible Navier-Stokes



# Place holder

Representative problem from:

“A Surrogate-Pareto Approach to Shape Optimization:  
Level Set Geometry Description” by John C. Otto  
Presented at the ICASE/LaRC Approximation Meeting  
July 21---23, 1997



# Illustrative Example (cont.)

## Preliminary Results on 1st Order Model Management

- **Preliminary results on first-order model management**
  - **Assumptions**
    - Model single periodicity cell (with doublets)
    - For initial computation, use fixed weights for the objectives
    - Use reduced problem ( 2 min. / N-S analysis)
  - **One function evaluation**
    - (at  $k=0$ , provide an initial point (3 or 6 variables))
    - Generate a grid
    - Input grid to the N-S code to generate the values of  $f_1$  and  $f_2$
  - **Other ...**
    - Derivatives are computed via finite differences
    - Lower-fidelity model is assumed to be a model with a coarser grid
  - **Preliminary impression**
    - Promising results for the chosen models



# Current Research: Constraints and MDO

- **Equality Constraints**

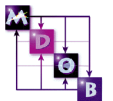
- Extension of the 1st order model management framework via the multilevel algorithm for equality constrained optimization (Alexandrov, 1993)
- Global convergence results (Alexandrov, 1997)

- **General NLP**

- Based on Alexandrov-El-Alem extension to general NLP of the multilevel algorithm for equality constrained optimization

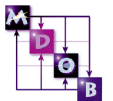
- **MDO**

- Many research questions, dependent on problem formulations



# Current Research: Some Novel Applications at MDOB

- **Applications**
  - High Speed Civil Transport (HSCT)
  - Aerospike Nozzle Design for RLV Concepts
  - Rotorcraft Blade Design
- **Common features:**
  - When used in high-fidelity mode
    - Large number of variables and constraints
    - Computationally expensive
    - Interest in using both statistical approximations and lower-fidelity physical approximations



# **Novel Applications: HPCCP HSCT**

(Weston, et al.)

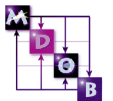
- **Problem Features:**

- **Components:**

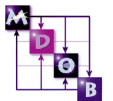
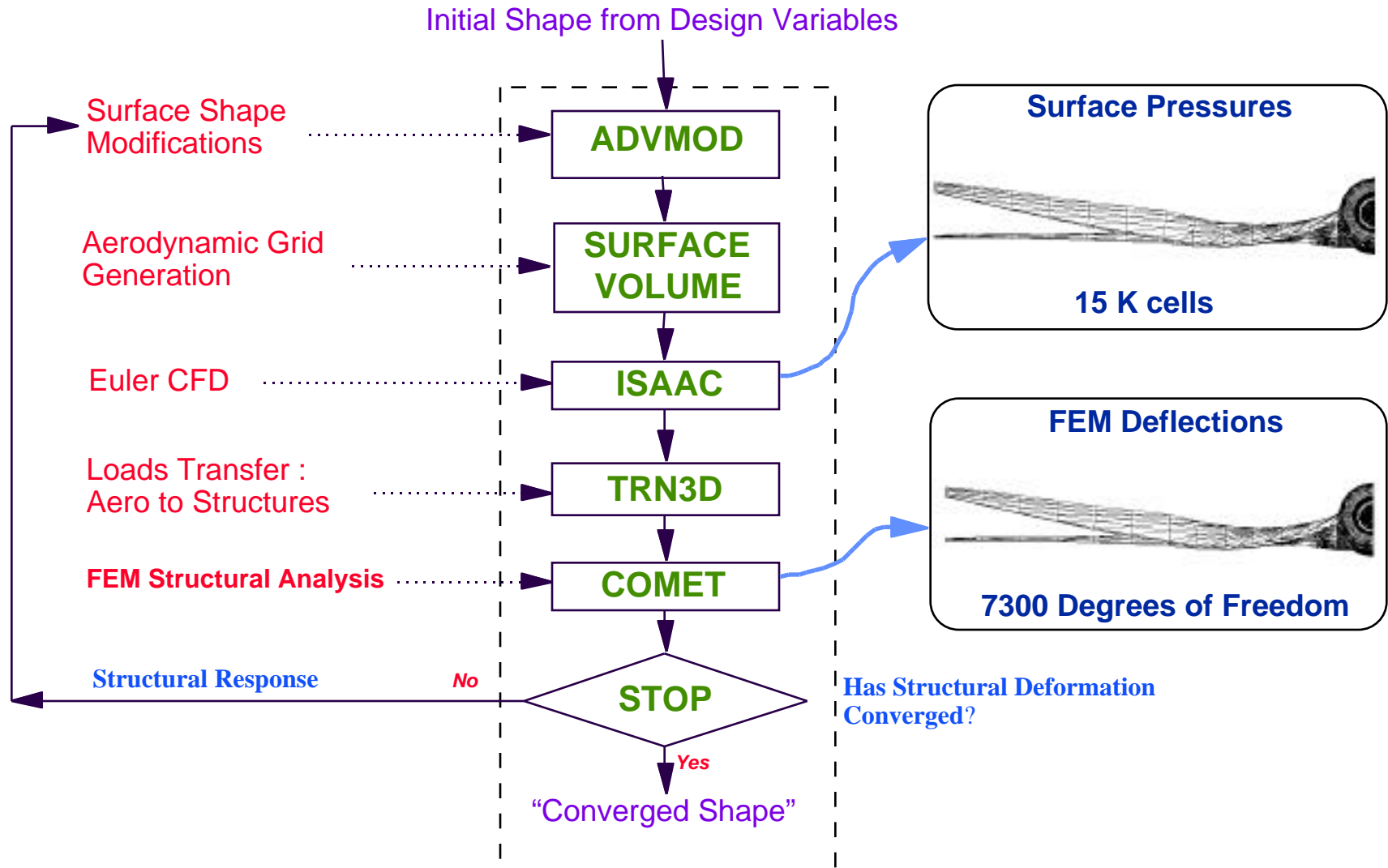
- **Multiblock Navier-Stokes CFD analysis & sensitivity**
    - **Adaptive FEM structural analysis & sensitivity**
    - **Many other disciplines**

- **Computationally intensive:**

- **Medium-fidelity**
      - **One aeroelastic function evaluation (multidisciplinary analysis of 5 Gauss-Seidel iterations) requires 6 hours on a heterogeneous network of 4-5 machines; 20 hours on a dedicated machine**
    - **High-fidelity**
      - **One aeroelastic function evaluation is expected to require 5-6 days on a dedicated machine; 2 days on a parallel machine; 3-6 hours on 64-processor machine ( $O(10^2)$  hours total)**



# Novel Applications: HSCT - Key Steps



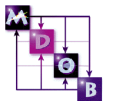
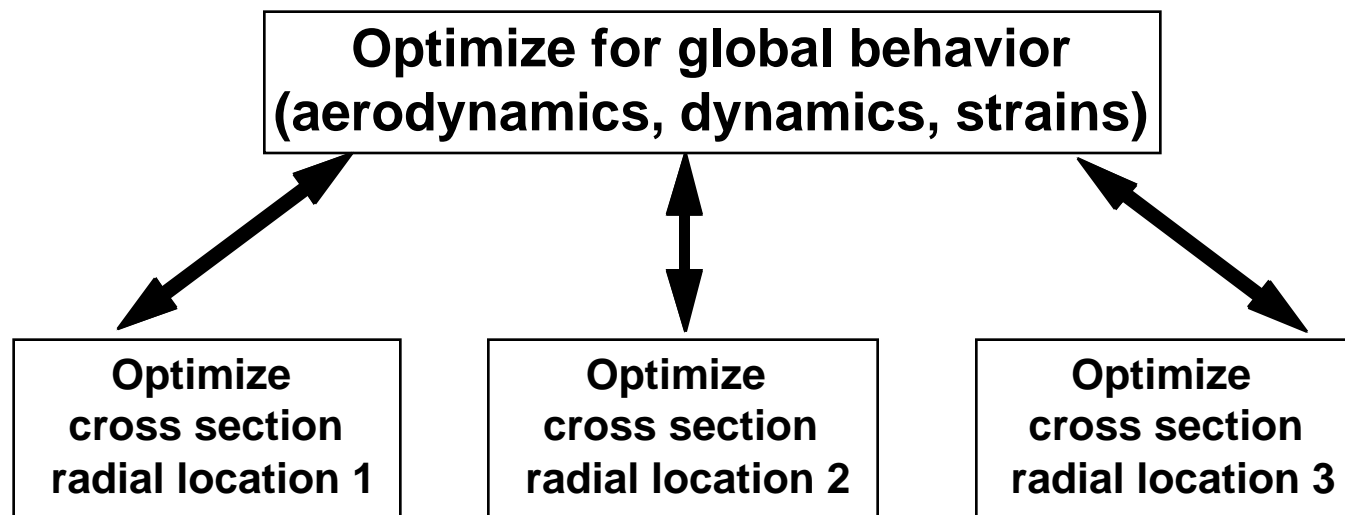
# Novel Applications: Rotorcraft Blade Optimization

(Walsh, et al.)

- **Problem Features:**

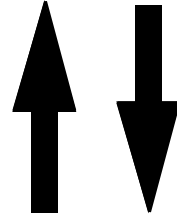
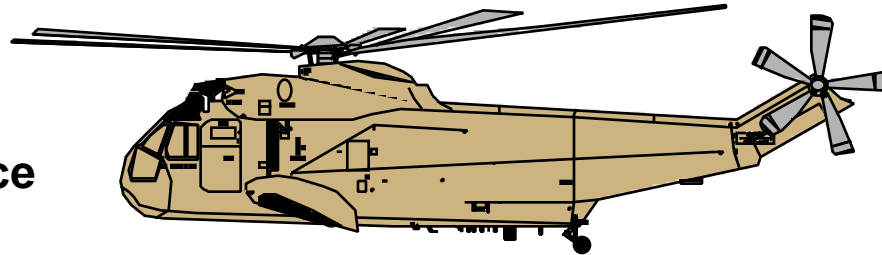
- Large number of design variables and constraints
- A multilevel approach to solution
- Computationally intensive
  - One function evaluation requires 30 minutes

## Integrated Aerodynamic/Dynamic/Structural (IADS) Solution Strategy



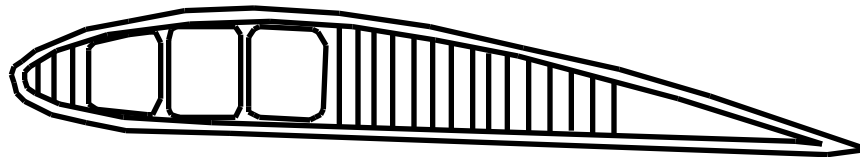
# Integrated Aerodynamics/Dynamics/Structures (IADS)

**UPPER LEVEL:**  
optimize performance  
and dynamics



Compromise between stiffness  
required by upper level and  
attainable in lower level

**LOWER LEVEL:**  
design internal  
structure at  
radial location  $i$





# **Novel Applications: Aerospike Engine Design**

(Korte, et al.)

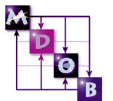
- **Problem Features:**

- **Components:**

- **Aerodynamics, structures, trajectory**
    - **High accuracy required due to sensitivity of SSTO vehicle performance**
    - **Minimize GLOW (Gross Lift-Off Weight) subject to structural constraints**
    - **One case: 16 variables, 596 structural constraints**
    - **Multidisciplinary feasible formulation used**

- **Computationally intensive**

- **Low-fidelity**
    - **Medium-Fidelity**
    - **High-fidelity**

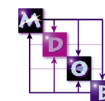


# RLV X-33 Concepts

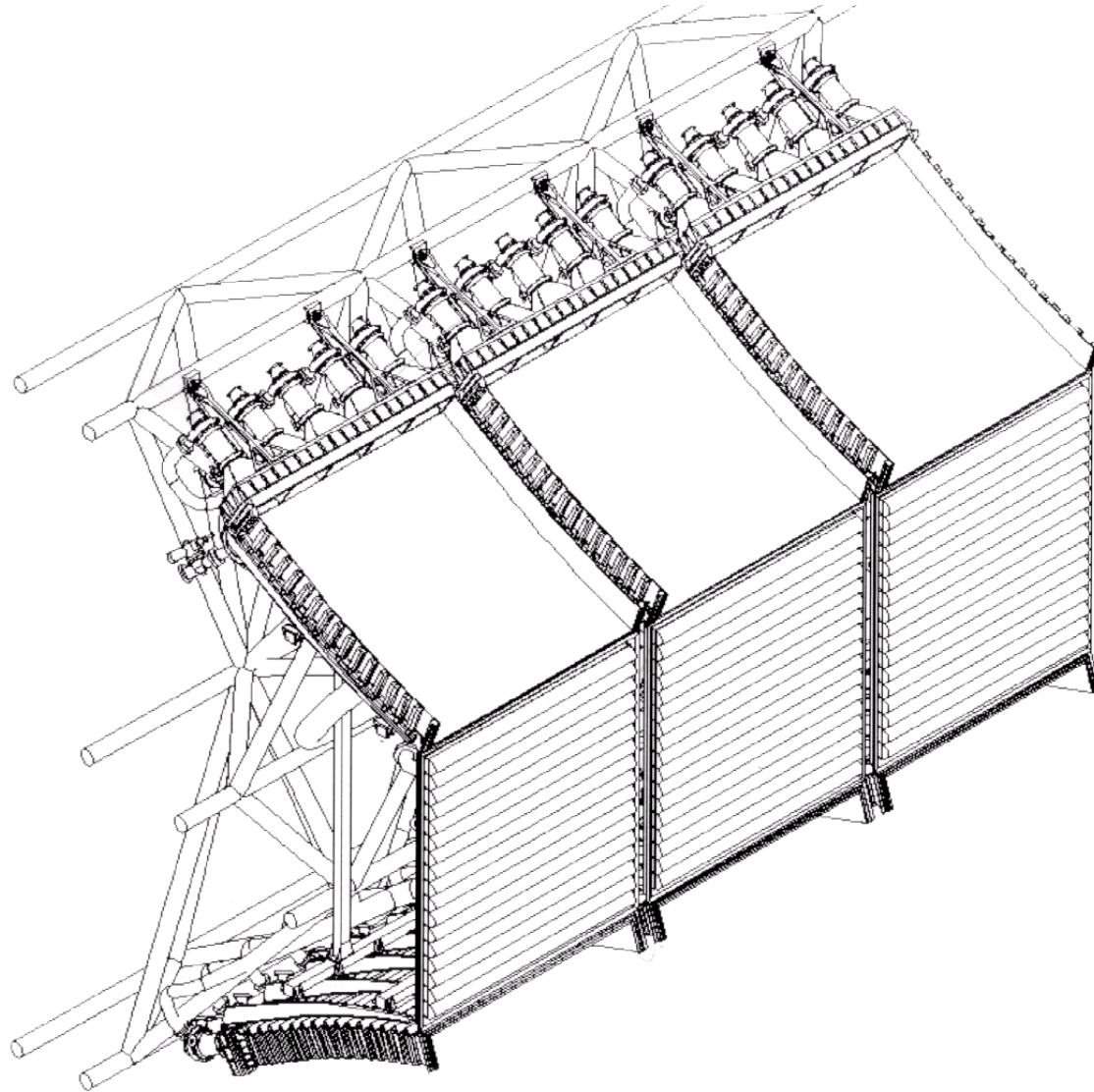
McDonnell Douglas/Boeing

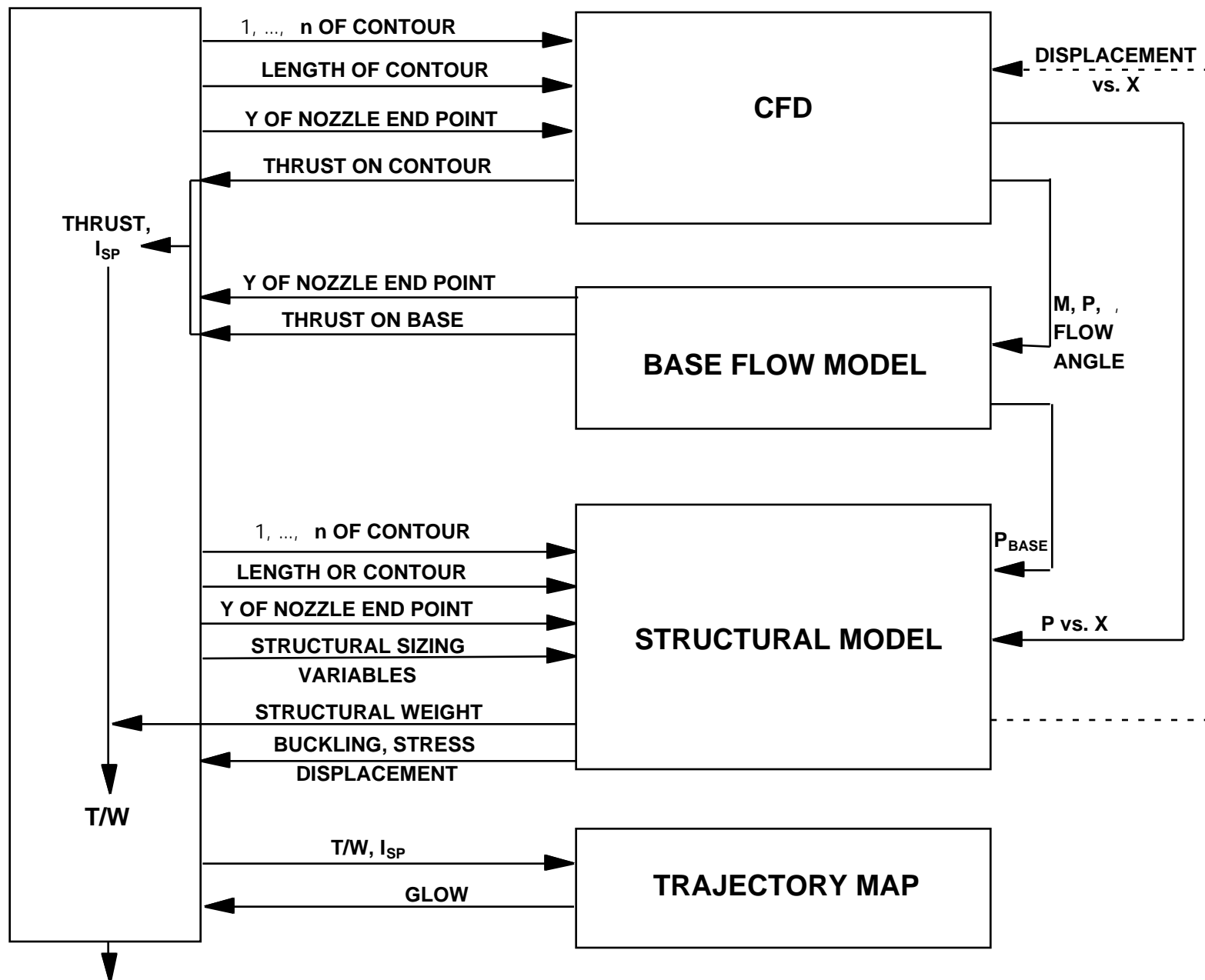
Lockheed Martin

Rockwell



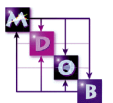
# AEROSPIKE ENGINE





MINIMIZE GLOW S.T. CONSTRAINTS

\*INCLUDES: HOT WALL THICKNESS, TUBE DIAMETERS, TUBE WALL THICKNESS, I-BEAM WEB THICKNESS, ET.



# Summary

- **Extensive, long-standing research on approximations in engineering optimization**
- **Benefit of research on approximations**
  - An opportunity to adapt state-of-the-art optimization algorithms to practical computational engineering
- **Introduced a framework for managing 1st order models in optimization**
  - Globally convergent
  - Arbitrary models with consistency requirements
- **Ongoing research - open questions:**
  - Usefulness in practice
    - Testing on increasingly realistic problems
    - Resolving consistency conditions in practice
  - MDO

